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**Information furnished in conformity with the Convention  
on Registration of Objects Launched into Outer Space****Note verbale dated 22 October 2025 from the Permanent Mission  
of the Republic of Korea to the United Nations (Vienna) addressed  
to the Secretary-General**

The Permanent Mission of the Republic of Korea to the United Nations (Vienna), in accordance with article IV of the Convention on Registration of Objects Launched into Outer Space (General Assembly resolution [3235 \(XXIX\)](#), annex), has the honour to transmit information concerning the space objects HYVRID, JACK-001 and ADDCUBE, launched by the Republic of Korea (see annex).<sup>1</sup>

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<sup>1</sup> The data on the space objects referenced in the annex were entered into the Register of Objects Launched into Outer Space on 23 October 2025.



## Annex

### Registration data on space objects launched by the Republic of Korea\*

#### **HYVRID**

|                                  |                                                                                                        |
|----------------------------------|--------------------------------------------------------------------------------------------------------|
| Name of space object             | HYVRID                                                                                                 |
| Name of launching State          | Republic of Korea                                                                                      |
| Date of launch                   | 15 March 2025 UTC                                                                                      |
| Location of launch               | Vandenberg Space Force Base, California,<br>United States of America                                   |
| Basic orbital parameters         |                                                                                                        |
| Nodal period                     | 95.78 minutes                                                                                          |
| Inclination                      | 97.40 degrees                                                                                          |
| Apogee                           | 590 kilometres                                                                                         |
| Perigee                          | 590 kilometres                                                                                         |
| General function of space object | Verification of Satellite Laser Ranging<br>(SLR) laser based precise orbit<br>determination technology |
| Other information                | Sun-synchronous orbit                                                                                  |

#### **JACK-001**

|                                  |                                                                                                                                                                                                                                                              |
|----------------------------------|--------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------------|
| Name of space object             | JACK-001                                                                                                                                                                                                                                                     |
| Name of launching State          | Republic of Korea                                                                                                                                                                                                                                            |
| Date of launch                   | 23 June 2025 UTC                                                                                                                                                                                                                                             |
| Location of launch               | Vandenberg Space Force Base, California,<br>United States                                                                                                                                                                                                    |
| Basic orbital parameters         |                                                                                                                                                                                                                                                              |
| Nodal period                     | 93.859 minutes                                                                                                                                                                                                                                               |
| Inclination                      | 97.438 degrees                                                                                                                                                                                                                                               |
| Apogee                           | 522.060 kilometres                                                                                                                                                                                                                                           |
| Perigee                          | 504.120 kilometres                                                                                                                                                                                                                                           |
| General function of space object | Verification of our internally developed<br>satellite working properly in the space<br><br>Image acquisition of Earth using an Earth<br>observation telescope. The image data will<br>be transmitted to ground stations via<br>radio-frequency communication |
| Other information                | 3U Cubesat<br><br>Sun-synchronous orbit                                                                                                                                                                                                                      |

\* The registration data are reproduced in the form in which they were received.

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**ADDCUBE**

|                                  |                                                           |
|----------------------------------|-----------------------------------------------------------|
| Name of space object             | ADDCUBE                                                   |
| Name of launching State          | Republic of Korea                                         |
| Date of launch                   | 23 June 2025 UTC                                          |
| Location of launch               | Vandenberg Space Force Base, California,<br>United States |
| Basic orbital parameters         |                                                           |
| Nodal period                     | 96.47 minutes                                             |
| Inclination                      | 97.746 degrees                                            |
| Apogee                           | 600.677 kilometres                                        |
| Perigee                          | 576.021 kilometres                                        |
| General function of space object | Satellite communications                                  |
| Other information                | Sun-synchronous orbit                                     |

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